Motivation & Problems
The first Moon robot weighed a ton. The Sojourner robot to Mars weighed only 11kg. As robots miniaturize there is always ambition to do more with less. This research investigates the correlation between size, mass, power consumption, and thermal consideration of small lunar rovers. The goals of this research is to:
1. Find the minimum size of lunar rovers
2. Investigate the effect of size and thermal coating on the operation time

Approach
We began by studying the characteristics of a lunar rover in development the Andy rover. The size, mass, and power consumption this rover was then proportionally scaled. The rovers radiator and electronics were discretized, then were simulated using finite element analysis in ANSYS. The rover’s scale was modified and the simulation repeated.

Investigated the minimum size of lunar rovers
A lunar rovers temperature is inversely proportional to its surface area. Smaller rovers have a much harder time cooling down. The minimum size of small lunar rover
- **1.95 kg:** a 17 °C margin was applied
- **1.6 kg:** an 11 °C margin was applied (with reliable data)
*45 and -20 °C are the critical operation temperatures.*

The coating effect on the operation time
Changing coating on the radiator can extend the operation time for around two days in the case of a 4 kg rover. However, the maximum temperature also increased, so the coating should be chosen carefully while considering the balance of the maximum and minimum temperature.

Conclusion
Based on the current temperature constraints of the components available the minimum mass of lunar rovers is determined as 1.95 kg. Also, we confirmed that the size of the radiator as well as its coating can change the operation time to some extent.

Future Work
Future work will need to incorporate radiation from the lunar surface. The mechanical designs of small lunar rovers will need to be altered to expand radiator area and to keep the electrical components close to any radiative surfaces. Therefore, after thermal experiments for calibrating this simulation, thermal and mechanical solutions should be integrated and the prototype should be manufactured in the near future.

Acknowledgements
Special thanks to the Planetary Robotics Laboratory and Dr. Whittaker for admitting me this lab, and Heather Jones, Daniel Arnett, and Oleg Sapunkov for giving me educational ideas, and the RISS program for this precious opportunity.

References